NASA/TM-2015-218672



The Growth of Small Corrosion Fatigue Cracks in Alloy 7075

Robert S. Piascik Langley Research Center, Hampton, Virginia

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Robert S. Piascik Langley Research Center, Hampton, Virginia

National Aeronautics and Space Administration

Langley Research Center Hampton, Virginia 23681-2199

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Nomenclature

Crack Length Silver/Silver Chloride Ag/AgCl da/dN Crack Growth Rate

ECT **Extended Compact Tension** Longitudinal-Transverse Sodium Chloride L-T

NaCl

K Crack-tip Stress Intensity Factor

Load P

Stress Ratio R

Abstract

The corrosion fatigue crack growth characteristics of small (>35 μ m) surface and corner cracks in aluminum alloy 7075 is established. The early stage of crack growth is studied by performing in situ long focal length microscope (500×) crack length measurements in laboratory air and 1% sodium chloride (NaCl) environments. To quantify the "small crack effect" in the corrosive environment, the corrosion fatigue crack propagation behavior of small cracks is compared to long through-the-thickness cracks grown under identical experimental conditions. In salt water, long crack constant K_{max} growth rates are similar to small crack da/dN.

1.0 Introduction

The propagation of small fatigue cracks from surface defects (5 μ m to 10 μ m) constitutes a large percentage (50% to 90%) of the total fatigue life of structural components [1]. Thus, accurate prediction of small crack da/dN is required for damage tolerant based life predictions. A problem arises because small cracks propagate at unpredictable accelerated rates compared to long fatigue cracks [2]. The difference in long and small crack growth behavior is due to numerous effects (i.e., diminished crack-tip shielding (closure) effects at small crack sizes). Compounding the small crack problem are complex environmental effects. Research on high strength steels in salt water has shown that chemically short (< 500- μ m) cracks propagate 1.5 to 500 times faster than long cracks subjected to the same mechanical driving force [3]. Limited data on 2000 series alloys exposed to deaerated salt water suggest little chemical-crack-length effect [4]. The purpose of this work is to study the chemical crack length effect in environmentally sensitive alloy 7075-T6.

2.0 Experimental Procedures

Constant amplitude fatigue crack growth experiments were conducted using a pin-loaded extended compact tension (ECT) specimen (also known as the eccentrically loaded single-edge cracked tension specimen) as shown in Fig. 1. The blunt notched alloy 7075-T6 sheet specimens were oriented in the longitudinal-transverse (L-T) direction. The growth of small surface and corner cracks located at the root of the polished (through 0.3-µm diamond paste) blunt notch were monitored *in situ* by using the long focal length (×500 magnification) microscope shown in Fig. 2. The surface crack length (a) was measured to an accuracy of approximately 3 µm in air and 5 µm in salt water. The acetate replica method [5] was also used to verify microscope-based surface dimension of small cracks at the root of the notch. Crack length measurements were conducted while the specimen remained at 70% of maximum load. Small surface and corner crack growth rates and stress intensity factors were calculated assuming uniform semicircular crack geometry and procedures documented elsewhere [4].

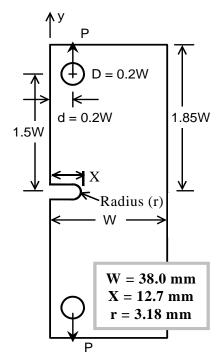


Figure 1. ECT specimen.

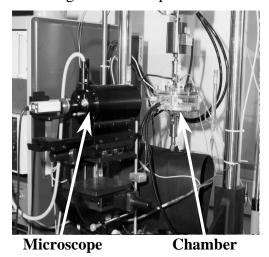


Figure 2. Corrosion fatigue setup.

Small crack experiments in laboratory air were conducted at constant ΔP and stress ratio (R) of 0.05 with a minimum crack length resolution of 35 μm . All corrosion fatigue tests were performed using the similar loading history; crack initiation was performed at constant ΔP (R = 0.05) and small crack growth testing was conducted at low and high R by varying P_{min} at constant P_{max} . Constant P_{max} testing eliminated possible load history effects when varying ΔK . Reduced *in situ* crack length measurement resolution limited the study of corrosion fatigue cracks to a ≥ 100 - μm and to high R (0.7 \leq R \leq 0.8) testing for the investigation of low ΔK environmental da/dN.

The direct current electrical potential drop method was used to monitor the growth of throughthe-thickness cracks. The computer-controlled long crack experiments were conducted by continuously adjusting the load to maintain the programmed ΔK and R. The decreasing ΔK test was conducted at variable R (0.2 \leq R \leq 0.09) at a constant K_{max} (15 MPa \sqrt{m}).

A 1.5-litre O-ring sealed plastic chamber, shown in Fig. 2, was used for salt-water corrosion fatigue experiments. The mid-half of the ECT specimen was continuously immersed in helium deaerated 1% NaCl (pH = 8) solution. All aqueous NaCl experiments were conducted at constant electrode potential using a potentiostat, two fully immersed platinum counter electrodes, and a silver/silver chloride (Ag/AgCl) reference electrode. All deaerated corrosion fatigue experiments were conducted at a constant electrochemical potential of –830 mV_{SCE} (approximately 100 mV_{SCE} cathodic of the breakaway or pitting potential). The distilled water/salt solution was continuously circulated through the test chamber at a rate of 30 ml/min. from a reservoir containing 25 liters of argon deaerated NaCl solution. A quartz window located on the side of the chamber and adjacent to the notch was used for *in situ* long focal length microscope viewing of the notch root surface fatigue cracks.

3.0 Experimental Results and Discussion

3.1 Fatigue Crack Growth Rate

Figure 3 shows the crack growth characteristics of small surface and corner cracks in air at R = 0.05. Here, data are shown for a range of crack lengths, $10 \, \mu m \le a \le 1000 \, \mu m$ for replicabased measurements and $35 \, \mu m \le a \le 1400 \, \mu m$ for microscope-based measurements. Typical of small crack data, the scatter is presumed to be a result of crack-front/microstructure interaction [2]. A power law fit of the microscope and replica data (solid and dotted lines, respectively) show that both measurement techniques yield nearly identical small crack da/dN for alloy 7075. Further comparisons with literature small crack data (dashed line power law fit) for similar low R (R = 0) show that the ECT results correlate well with replica-based data from single edge notch tests [6]. Long crack growth characteristics (R = 0) are similar to small crack da/dN for $\Delta K > 3 \, MPa\sqrt{m}$, but small cracks exhibit accelerated da/dN in the long crack growth threshold regime.

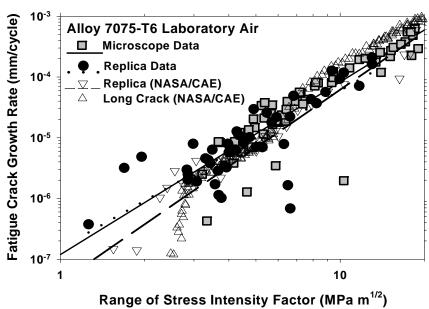


Figure 3. The fatigue crack growth characteristics of small cracks in air.

The results in Fig. 4 show that small cracks ranging in size from 100 μ m to 1400 μ m exhibit accelerated fatigue crack growth rates in deaerated 1% NaCl compared to laboratory air at a loading frequency of 5 Hz. A comparison of the salt water and air power law fit (solid and dashed lines, respectively) show that fatigue crack growth rates are accelerated by a factor of three in NaCl. The corrosion fatigue crack propagation behavior of small cracks is also compared to long crack da/dN shown in Fig. 4. Long crack growth rates at a constant $K_{max} = 15 \text{ MPa}\sqrt{m}$ are essentially closure free, accelerated compared to R = 0.1 da/dN, and envelop the majority of small crack data; similarly, small and long crack behavior in deaerated salt-water suggests no crack length chemical effect. The increase in long crack constant K_{max} and R da/dN for $\Delta K > 7 \text{ MPa}\sqrt{m}$ may be a result of stress corrosion cracking effects.

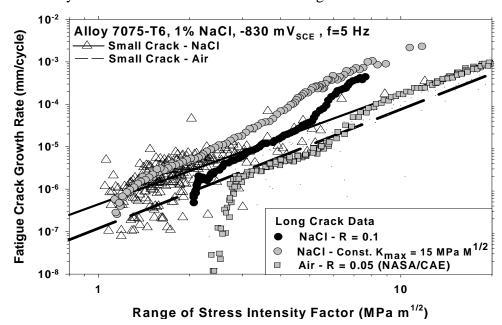


Figure 4. The corrosion fatigue crack growth characteristics of small and long cracks in deaerated salt water.

3.2 Small Crack Fractography

Figure 5 shows the region of crack initiation and the typical morphology of a semicircular-shaped corrosion fatigue surface crack (a = 355 μ m) that was exposed to deaerated NaCl (-800 mV). The micrographs are oriented normal to the fatigue crack surface and the surface of the blunt notch is located at the bottom of both micrographs. The fatigue crack initiated at a corrosion pit located at the root of the blunt notch (arrow shown in Fig. 5a). Following initiation, the crack propagated along a transgranular semicircular-shaped crack path marked by the dashed lines in Fig. 5a. The high magnification micrograph in Fig. 5b shows that the fatigue crack initiated at a corrosion pit approximately 10 μ m in depth. The pit is elongated in shape with microcrack-like features emanating from the region of greatest pit depth. Figure 6 shows typical examples pit surface morphology. The micrographs in Fig. 6 are oriented normal to the corroded blunt notch surface and in the highly stressed region very near the fatigue crack. Here, both micrographs show that surface pits, 10 μ m to 20 μ m in size, contain regions of microcracking. The pit shown in Fig. 6b exhibits a microcrack-like morphology similar to the subsurface pit morphology observed in Fig. 5b.

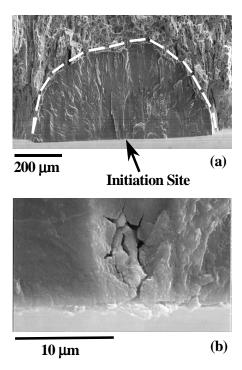


Figure 5. Fatigue crack/initiation.

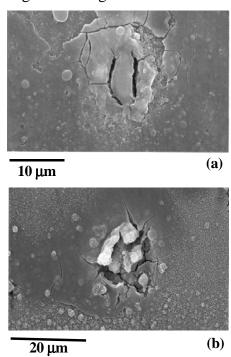


Figure 6. Surface pitting.

4.0 Conclusions

No chemical crack length effect was observed. For deaerated salt-water solution, small surface and corner fatigue cracks exhibited similar crack growth rates as those observed for long thoughthe-thickness cracks. Small cracks exposed to salt water exhibited a factor of three increase in

da/dN compared to small crack growth rates in air. A similar factor of three increase in long crack da/dN (R = 0.1) was observed in NaCl compared to long crack growth rates (R = 0.05) in air. Presumably, small crack growth rate data shown herein was truncated near a $\Delta K = 1$ MPa \sqrt{m} due to crack length resolution. It is likely that small cracks grow at accelerated rates for $\Delta K < 1$ MPa \sqrt{m} .

5.0 References

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REPORT DOCUMENTATION PAGE

Form Approved OMB No. 0704-0188

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4. TITLE AND SUBTITLE 5a. CC			NTRACT NUMBER
The Growth of Small Corrosion F	atigue Cracks in Alloy 7075		
		5b. GR	ANT NUMBER
		5c. PR	OGRAM ELEMENT NUMBER
6. AUTHOR(S)		5d. PR	OJECT NUMBER
Piascik, Robert S.			
		5e. TA	SK NUMBER
		5f. WO	RK UNIT NUMBER
		86902	1.03.07.01.07
7. PERFORMING ORGANIZATION	NAME(S) AND ADDRESS(ES)		8. PERFORMING ORGANIZATION REPORT NUMBER
NASA Langley Research Center Hampton, VA 23681-2199			
Trampion, VA 23081-2199			L-20516
9. SPONSORING/MONITORING AG	ENCY NAME(S) AND ADDRESS(ES)		10. SPONSOR/MONITOR'S ACRONYM(S)
National Aeronautics and Space A Washington, DC 20546-0001	dministration		NASA
			11. SPONSOR/MONITOR'S REPORT NUMBER(S)
			NASA/TM-2015-218672
12. DISTRIBUTION/AVAILABILITY S	TATEMENT		

Unclassified - Unlimited

Subject Category

Availability: NASA CASI (443) 757-5802

13. SUPPLEMENTARY NOTES

14. ABSTRACT

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15. SUBJECT TERMS

Fatigue Crack Growth; Corrosion; Fractography; Extended Compact Tension; Aluminum Alloy

16. SECURITY CLASSIFICATION OF:		17. LIMITATION OF ABSTRACT	18. NUMBER OF	19a. NAME OF RESPONSIBLE PERSON	
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